

and having an annular outwardly facing channel disposed therein, said inner ends of said exit guide vanes including tangs extending radially inwardly into said channel, the width of said channel in the axial direction being wider than the thickness of said tangs in the axial direction, permitting axial movement of the inner ends of said vanes, said inner ends of said exit guide vanes being free to move radially relative to said fairing means.

2. The turbine exhaust case assembly according to claim 1 wherein said vanes each have a forward and rearward edge and are attached to said outer case means adjacent their forward edges.

3. The turbine exhaust case assembly according to claims 1 or 2 wherein said wall means includes axial corrugations between said struts.

4. The turbine exhaust assembly according to claim 1 wherein said tangs are circumferentially extending and abut one another to define a segmented annular ring.

5. The turbine exhaust case assembly according to claim 1 wherein said engine includes engine mount ring means disposed radially outwardly of said outer case means, an axially extending rotor shaft, bearing means disposed on said shaft, and bearing support structure secured to said bearing means and located radially inwardly of said inner case means, said inner case means being attached to and supported by said bearing support structure and said outer case means being supported by said inner case means through said struts, a plurality of substantially radially extending tie rods each passing through one of said struts, each tie rod having an inner end attached to said bearing support structure and an outer end attached to said mount ring means to transmit loads from said support structure to said mount ring means.

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